

**GSFC** · 2015

# Some General Principles in Cryogenic Design, Implementation, and Testing

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#### **Outline**

- Opening remarks
- The role of thermodynamics
- General design principles
- Properties of materials
- Producing "cold"
- Cryo-cooling in space
- Instrumentation
- Heat switches
- Superconductivity
- Cooling Below 1 K

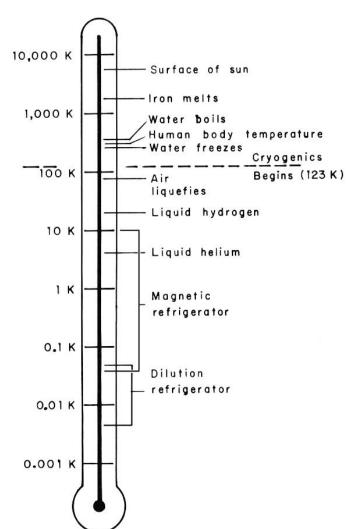


# What is "Cryogenic"

- For the purposes of this talk, T < 100 K is cryogenic</li>
  - Air liquefies
  - Certain metals and ceramics become superconducting
  - Out of the realm of our normal experience (arctic conditions are not cryogenic)
  - Heat capacities decrease from the Dulong & Petit (3/2 R) value
  - In general the physics becomes different from room temperature



# The Logarithmic Temperature Scale



- Note use of absolute scale
- Each decade corresponds to different physics and different solutions to design problems
  - 100-1000 K is the range we are used to
  - 10-100 K, air liquefies and solidifies, High temperature superconductivity
  - 1-10 K, low temperature superconductivity, liquid helium
- Note that properties are not "constant" any more, so concepts like "average" temperature must take this into account
- When analyzing a system, heat flow margin will take the place of temperature margin



# Thermomdynamics is a Serious Subject!



Robert Boyle 1627-1691



Benjamin Thompson Count Rumford 1753-1814



Nicolas Léonard Sadi Carnot 1796-1832



J. Willard Gibbs 1839-1903



Heike Kamerlingh Onnes 1853-1926



Max Planck 1858-1947



James P. Joule 1818-1889



Rudolf Clauxius 1822-1888



Gustav Robert Kirchhoff 1824-1887



Walther Nernst 1864-1941



Constantin Carathéodory 1873-1950



Albert Einstein 1879-1955



William Thomson Lord Kelvin 1824-1907



Clerk Maxwell 1831-1879



Peter Debye 1884-1966



F. E. Simon 1893-1956



# The Laws of Thermodynamics

- First Law of Thermodynamics (Conservation of Energy)
  - Energy in = Work out
  - you can't get something for nothing
- Second Law of Thermodynamics (Entropy)
  - $\partial$ Entropy ≥ ( $\partial$ Energy/Temperature)
  - you can't break even
- Third Law of Thermodynamics (Absolute Zero)
  - Entropy -> 0 as Absolute Temperature -> 0
  - there's no use trying



# **Thermodynamics**

- Thermodynamics is key to understanding cryogenic processes
- Refrigeration
  - 1st and 2nd laws of thermodynamics
- Approach to Absolute Zero
  - 3<sup>rd</sup> law of thermodynamics



# **Staging**

- Intercept heat in stages to reject heat at the highest possible temperature
- In general heat rejection difficulty goes as T<sup>-2</sup>



# Design: The "KISS" Principle

- Start with a design that can be calculated using "back of the envelope" methods
  - Make all components easy to analyze
    - Analysis effort should not be underestimated!
  - The fewer items that are crucial in a design the better
    - Simpler analysis
    - Simpler construction
    - Simpler validation



# **Example**

- GSE motor driven photogrammetry cameras for JWST
  - Original concept: camera housing to cool passively through incidental contact in motor and gears
    - Very difficult to model and verify performance
    - Became an extra potential heat source that had to be tracked
  - Solution: make system "deterministic" by using thermal straps



# **Estimating by Previous Example**

- Previous systems have been made that can be used as a "jumping off" point for a thermal design or estimating cooling requirements
  - Various parametrizations have been used to give an analog expression to these extrapolations or interpolations of actual systems



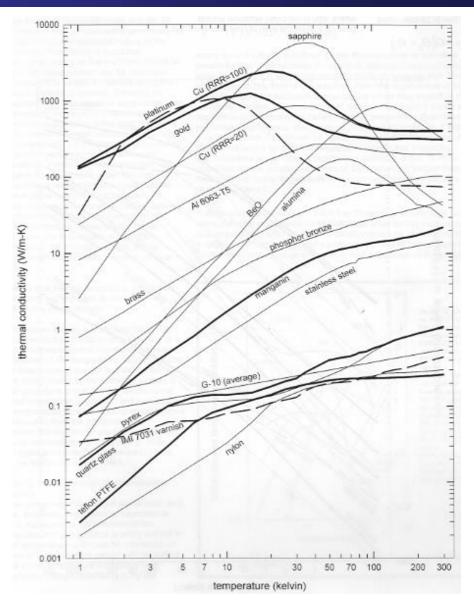
# **Properties of Materials**

- Thermal Conductivity
- Thermal Absorptivity and Emissivity
- Strength and Brittleness Properties
- Electrical Conductivity
- Specific Heat
- Gases and Liquids (density & pressure vs. temperature, heat of vaporization and melting, crystal structure, etc.)
  - Example: solid nitrogen has a low temperature change of phase which causes an expansion. This was learned by NICMOS at the cost of a compromised mission



# **Conductivity Graph**

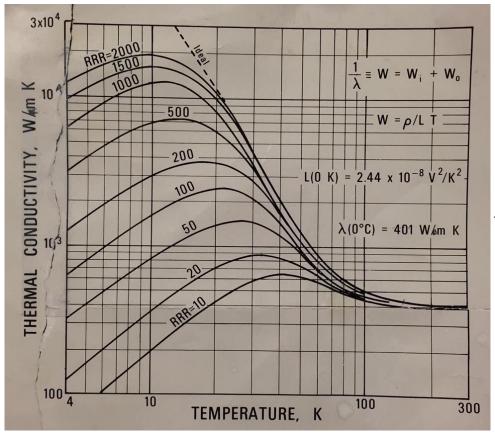
 Thermal conductivity varies greatly between room T and low T





# **High Purity Metals**

 At low temperature electrons have fewer phonons to scatter from, so the thermal conductivity goes up until defects and impurities dominate



RRR = residual resistance ratio
A measure of the purity of the metal
And its crystallinity



#### Wiedemann-Franz

- Electrons carry the heat in metals
- W-F is a relation between electrical and thermal conductivity

```
\rho = L_0T/K
```

Where  $\rho$  = resistivity, T = absolute temperature, K = thermal conductivity, and L<sub>0</sub> = Lorentz constant = 2.44 x 10<sup>-8</sup> V<sup>2</sup>/K<sup>2</sup>

Not applicable to superconductors

#### Emissivity and Absorptivity: Temp. and Wavelength Dependence

- The emissivity of most materials is temperature and wavelength dependent
  - Requires wavelength dependent analysis for radiation which is usually accomplished by creating a few wavelength bands in the analysis software
  - Experience on JWST shows that 3 wavelength bands representing the major "hot" (60 K < "hot" < 300 K) sources provide enough accuracy without greatly increasing model run time



## **Properties of MLI**

The Lockheed Equation

$$Q/A = [(C_sN^{3.56}T_m)/(N_s+1)](T_h-T_c) + [(C_r\varepsilon_{tr})/N_s](T_h^{4.67}-T_c^{4.67})$$

Where Q/A is W/m<sup>2</sup>, Tm is the average of  $T_h$  and  $T_c$ , N is the layer density in layers per cm,  $N_s$  is the total number of layers,  $\varepsilon_{tr}$  is the surface emissivity,  $C_s$  is 2.11e-9, and  $C_r$  is 5.39e-10.

- Degradation of MLI at lower T
  - Basically dominated by thru-layer conduction at low T
- Structural MLI
  - Each layer is separated by a well defined spacing and has some structural qualities
- Lateral conduction
  - May be minimized by slitting



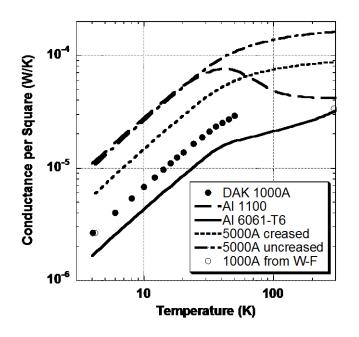
# **DAK Emissivity vs. T**

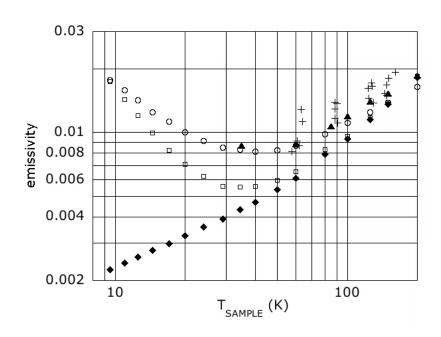
Metals' emissivity follow the Hagen-Reubens relation to first order:

R ~ 1-2[(2
$$\epsilon$$
0 $\omega$ )/ $\sigma$ ]0.5

R is the reflectivity,  $\epsilon 0$  is the permittivity of vacuum,  $\omega$  is the frequency of the radiation and  $\sigma$  is the conductivity of the metal surface

But DAK's metal is thin







# **Suitable Materials for Cryo**

- Austenitic stainless steels: 304, 304L, 316, 321, A286
- Aluminum alloys: 6061, 6063, 5083, 2219, 1100
- Copper: OFHC, ETP and phosphorous deoxidized
- Brass
- Fiber reinforced plastics: G –10 and G –11, CFRP
- Niobium & Titanium (frequently used in superconducting RF systems)
- Invar (Ni /Fe alloy)
- Indium (used as an O ring material)
- Kapton and Mylar (used in Multilayer Insulation and as electrical insulation
- Teflon (does not become brittle, but creeps)
- Quartz (used in windows)



# **Unsuitable Materials for Cryo**

- Martensitic stainless steels Undergoes ductile to brittle transition when cooled down.
- Cast Iron also becomes brittle
- Carbon steels also become brittle. Sometimes used in 300 K vacuum vessels but care must be taken that breaks in cryogenic lines do not cause the vacuum vessels to cool down and fail
- Rubber and most plastics
  - Plastic insulated wires are frequently OK as long as the wire is not repeatedly flexed which could lead to cracking of the insulation (check outgassing first)



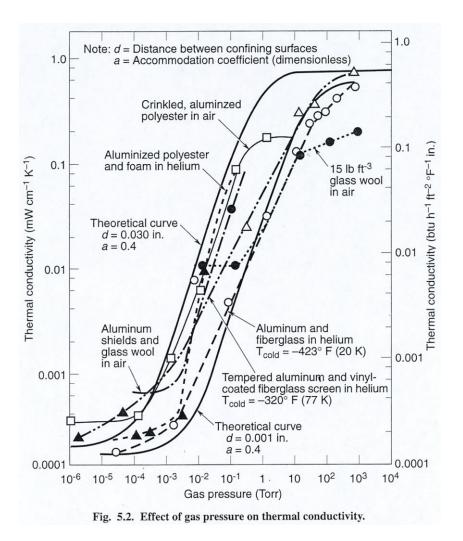
#### **Gas Conduction-1**

#### Can be bad

- Unwanted thermal shorts in a test
- Failure of the XRS instrument on Astro-E2

## Can be good

- Gas gap heat switches
- Aid to speed cool down and warm up
- Can be used as a substitute for a failed heat switch





#### **Gas Conduction-2**

#### Molecular Heat Transfer

- Gas density is lower than mean free path between objects
- Heat transfer depends on the temperature difference but not on the separation distance

#### JWST example

- To shorten the cool down time from room T (300 K) to 30 K
   helium exchange gas is used (~10<sup>-2</sup> Pa) within the chamber
  - Mean free path is exceeded for 10<sup>-3</sup> to 100 Pa depending on objects' spacing

#### ASTRO-H example

- In the EM dewar a heat switch failed open
- To operate the adiabatic demagnetization refrigerator, a method to use ~10<sup>-3</sup> Pa of gaseous helium to remove the heat of magnetization was used successfully
  - Too much gas causes excessive thermal shorting to warmer components in the dewar



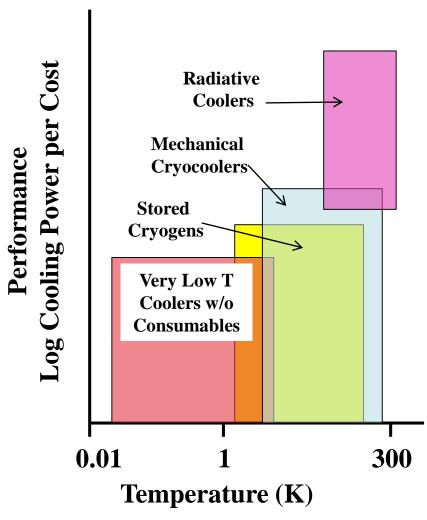
# **Producing Cold: Cryogens and Cryocoolers**

- Mechanical cryocoolers
- LN<sub>2</sub>, LHe, etc.
- Supplement through use of intermediate cooling stages
  - Vapor cooling
  - radiators



# **Producing Low Temperatures in Space**

Radiation can only work to ~ 30 K, practically



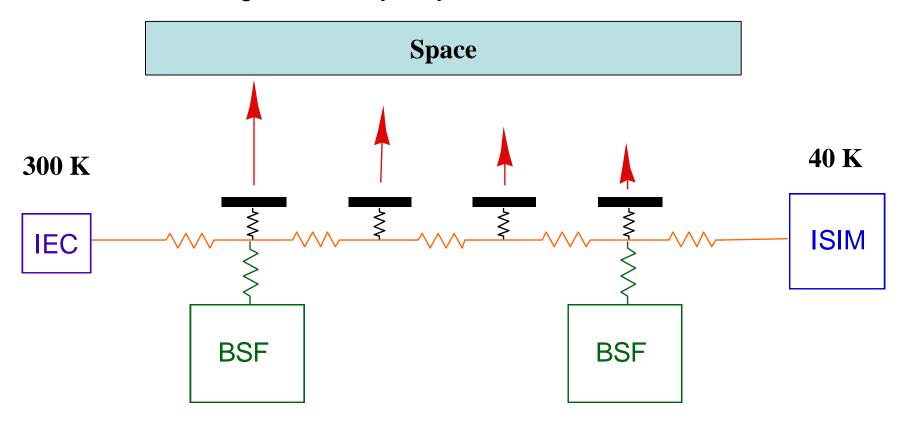


# Radiators in Space

- Some flight heritage at cryogenic temperatures (COBE, Landsat, Cassini/CIRS, MAP, Spitzer)
  - JWST will use radiative cooling
  - Successful test of Subscale Cryo-optical Thermal Testbed in support of ST-9 Large Space Telescope proposal
- Operate from room temperature (and above) to as low as 30 K
  - Depends strongly on mission design
- Passive heat rejection
  - Sunshade/earthshade provides shielding from incoming radiation
  - Radiator with a view of deep space connects to heat source (instrument, optics, part of spacecraft bus) by means of a thermal distribution system
    - Metal conductors
    - Loop heat pipes
  - Requires heaters/thermostats to regulate temperature
- Require stringent controls to meeting thermal budgets
- Spitzer reached 34 K on radiative outer shell
- JWST expects to reach ~26 K on instrument radiators

# Different Geometry - JWST Harness Radiator

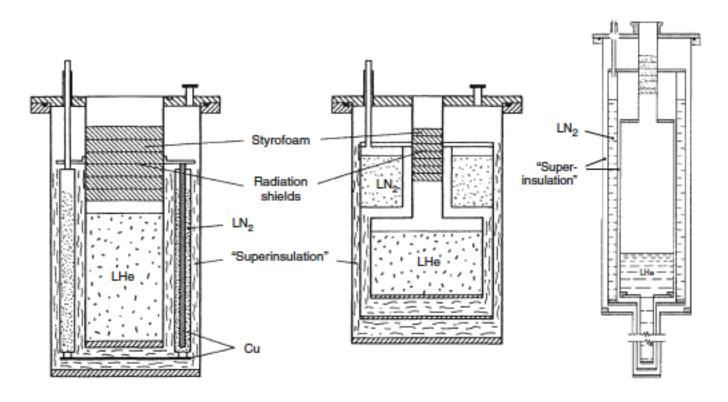
- 4 stages of isolation to limit the amount of heat flowing from the room temperature electronics (IEC) to the cold instruments (ISIM)
  - Coldest stage is actually only isolator





#### **Dewar Construction**

- Dewar must trade mechanical robustness with thermal isolation
- For liquid helium dewars usually have conduction and radiation heat loads roughly the same
- Use vapor cooled shields to intercept heat at higher temperatures





# **Working with Cryogenic Fluids**

#### In general:

- Low heat of vaporization
- Can be pumped or pressurized to change boiling point
- Can freeze if too cold (except helium)
- Low to zero contact angle, i.e., wets all surfaces
- Represents a large potential energy in a sealed container

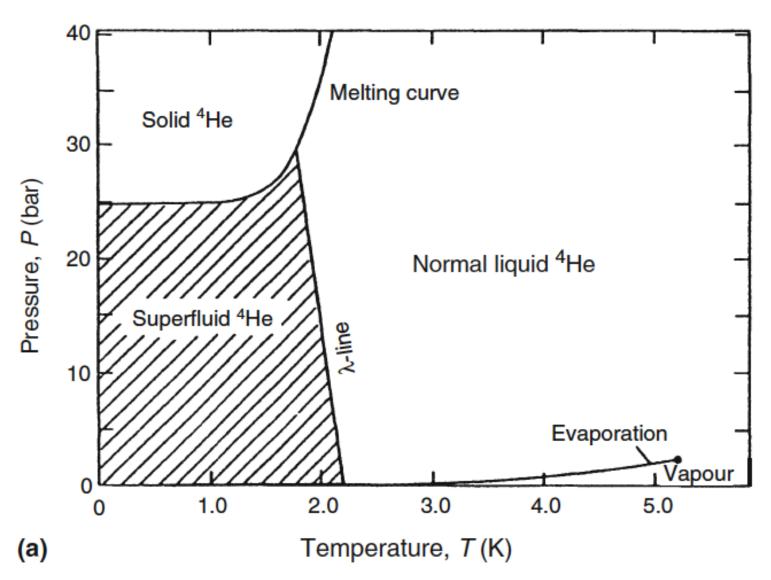
IRAS Dewar Launched 1983
First Superfluid He Dewar in Space
(not as large as it appears)



TFAWS 2015 - August 3-7, 2015



# The Unique Phase Diagram of <sup>4</sup>He





# Lab Cooler - Gifford McMahon Cycle

- Gifford-McMahon Refrigeration Cycle
  - Regenerator stores heat in compression phase, and releases heat in expansion phase
  - Compress while most of the gas is at warm end, and expand while most of the gas is at the cold end
  - Reverse the phase, and you have an expensive heater!





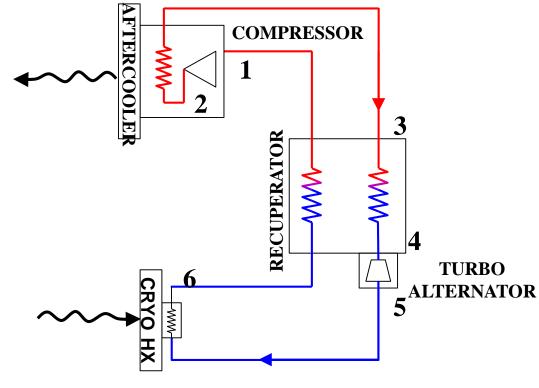
# **Cryocoolers for Space Use**

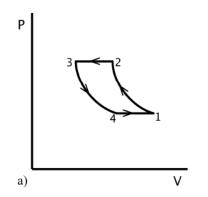
- Mass, size, input power, and reliability are drivers
- Reverse Brayton Cycle
- Stirling Cycle
- Pulse Tubes
- Joule/Thomson Coolers

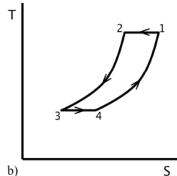


# **Reverse Brayton Cycle**

Turbo alternator removes work from cold stage, therefore increasing cooling

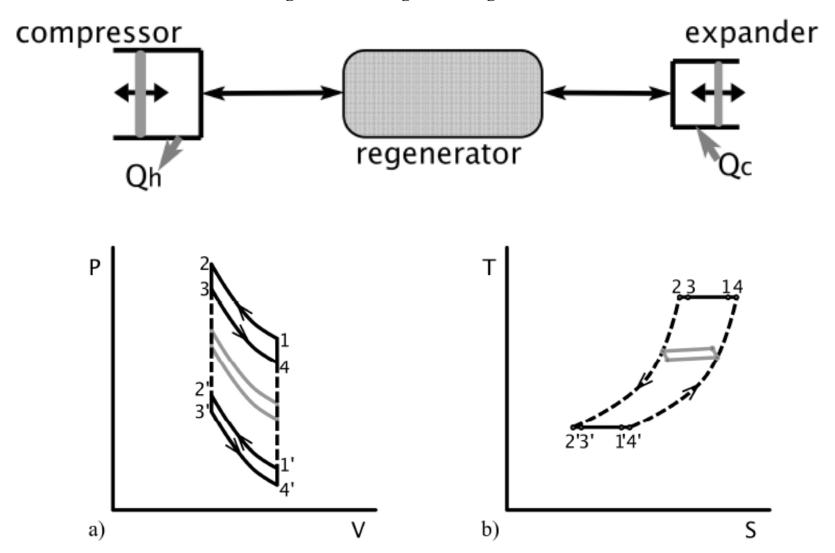






# Regenerator Cycle (Stirling and Pulse Tube)

From P. Kittel, "Are P-V and T-S Diagrams Meaningful for Regenerative Coolers?"





# **Stirling Cycle**

- Similar to GM cycle
  - Identical function of regenerator in coldfinger
  - Pressure cycle driven by oscillator rather than tanks, valves and a compressor
  - Phase angle controlled electrically, mechanically, or pneumatically
- Easier to miniaturize than GM



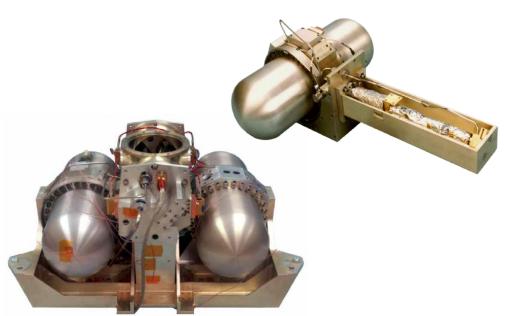




#### **Pulse Tubes**

#### Similar to Stirling cycle

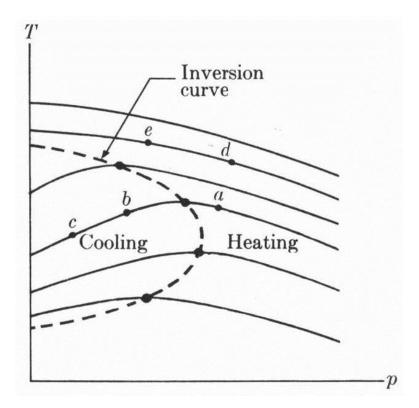
- Identical function of regenerator in coldfinger, pressure cycle driven by oscillator
- Phase angle controlled by resonant gas volume
- Simpler <u>mechanism</u> than Stirling, but a whole new set of gascontrol challenges





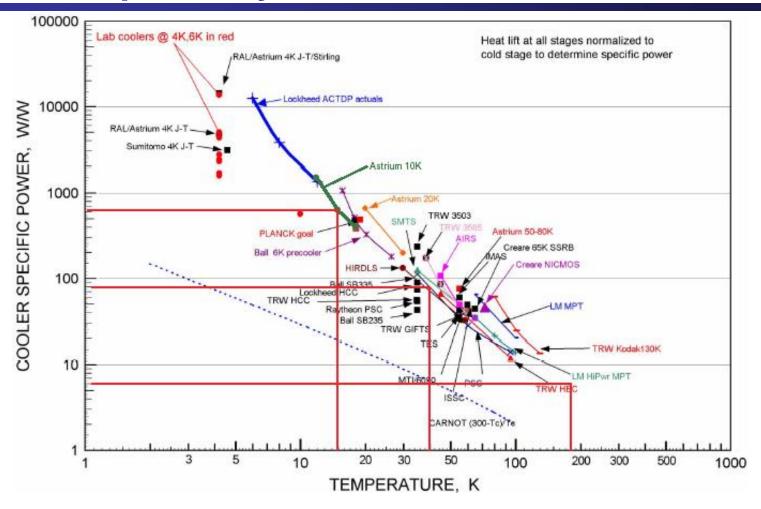
# **Joule Thomson Expansion**

 Gas must be precooled and not too high in pressure to produce cooling when expanded isenthalpically





## **Space Cryocooler Performance**



 Roughly T<sup>-2</sup> dependence on input power to cooling power ratio



#### Instrumentation and What is Important to Measure

- Thermometry, thermometry, thermometry
- Pressure for fluids
  - May be in situ or reading vapor pressure
- Pressure for vacuum
  - Pressure reading depends on temperature
    - $P_A = P_B (T_A/T_B)^{1/2}$
  - For example a pressure gauge on the vacuum wall of a thermal/vacuum chamber will read higher than the actual pressure in a cold shroud



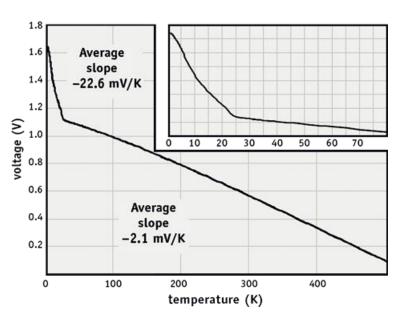
#### **Thermometry**

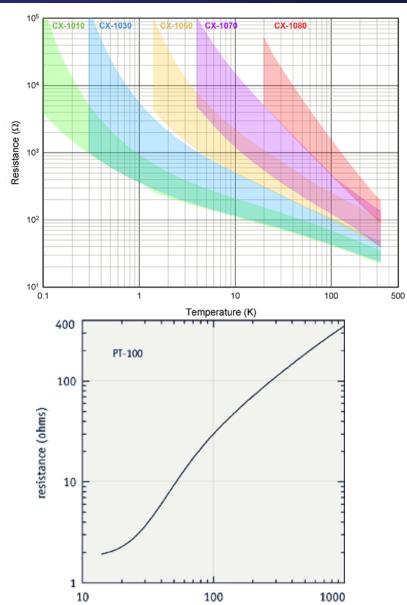
- Select thermometer type based on temperature range
  - Use 4 lead devices where high sensitivity and accuracy are required
    - Remove thermal emfs by reversing current
      - Not possible with diodes
- Self heating can produce erroneous readings in thermistors
  - Function of power and temperature
    - Readout power applied = 10<sup>-9</sup> T<sup>2</sup>
  - But, higher voltage can be used to obtain higher sensitivity at a cost of accuracy



## **Thermometry Chart**

- Figure of merit for thermisters is 1/R•dR/dT
- Cernox best < 70K</li>
- Pt best for > 70 K
- Si diodes good over wide range
- Thermocouples have very poor sensitivity below 100 K



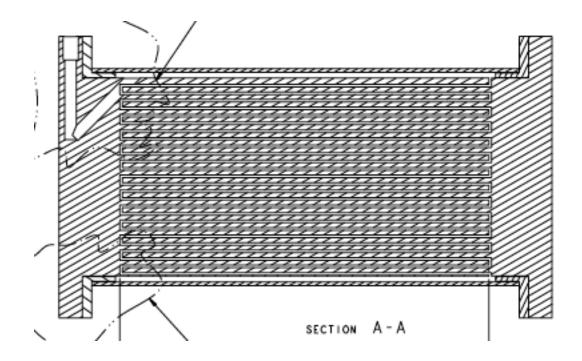


temperature (K)



#### **Heat Switches-Gas Gap**

- Uses closely spaced conductors separated by thermally insulating material
- Gas is admitted to gap or pumped out by heating or cooling an adsorbing material (getter)

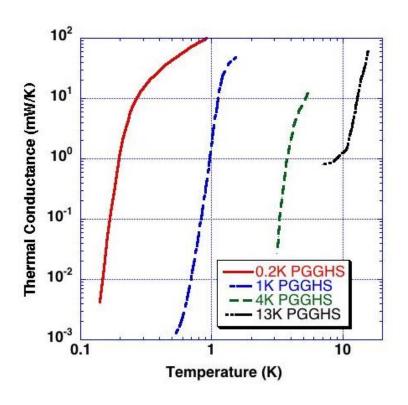




#### **Heat Switches-Passive Gas Gap**

- Passively Operated Gas Gap Heat Switch
  - Getter is thermally attached to the normally cold end







#### **Heat Switches-Mechanical**

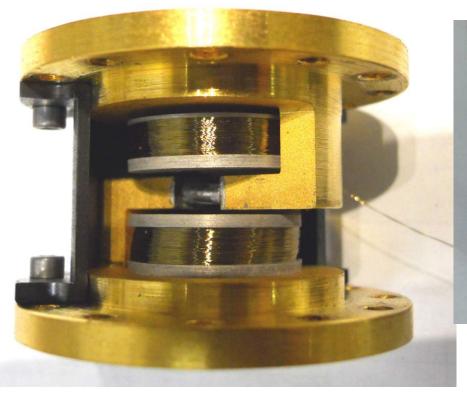
- Differential contraction
- Motor driven
- Manual
- Magnetostrictive
- Piezoelectric



### **Heat Switches - Other**

#### **Superconducting**

#### Magnetoresistive







#### Superconductivity

- Quantum mechanical effect where electrons in certain conductors combine to form "Cooper pairs"
  - Transition point affected by temperature, current density, and magnetic field
- Characterized by zero electrical resistance and drop in thermal conductivity
  - Cooper pairs carry current and pass through the material without interacting
  - May be used for low T heat switch
- Types of superconductors
  - Type I Generally pure metals, T<sub>c</sub> < 10 K</li>
    - Also can be used as a magnetic shield
  - Type II Alloys, some pure metals,  $T_c$  < 20 K
    - Can remain superconducting in higher fields
  - MgB<sub>2</sub> Magnesium Diboride, T<sub>c</sub> ~39 K
  - High Temperature Superconductors (HTS) Ceramics, T<sub>c</sub> < 110 K</li>



### **High Temperature Superconductivity**

- Usually a ceramic consisting of RBCO, where R is a rare earth element, for instance YBCO, yttrium barium copper oxide
- Can make large/high field coils
- Joints have small amount of resistance so coil is not "persistent"
- Best performance is for bulk or flat tapes made with a thin film deposition
  - Round wire forms are now being explored



# **Making Use of Superconductivity**

- i<sup>2</sup>R-free coils for motors and actuators
  - Also proposed for energy storage
- Low thermal conductance high current wiring
- SQUIDs (Superconducting Quantum Interference Devices)

#### Magnet that produces 3 T with 2 A input





### In the Regime of Sub Kelvin Temperatures

#### Quantum behavior

- <sup>3</sup>He has Fermi-Dirac statistics (like electrons) and <sup>4</sup>He has Bose-Einstein statistics (like photons)
- Helium does not freeze at atm. pressure

#### • <sup>3</sup>He and <sup>4</sup>He

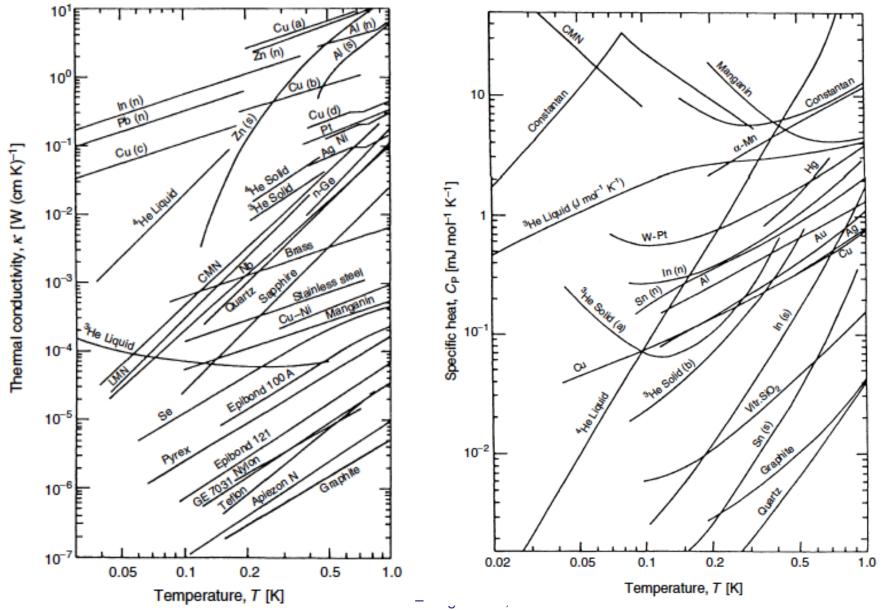
3He dissolves in 4He, creating an opportunity for cooling

#### Boundary resistance

- Not thermal contact per se, but a thermal conductance that depends only on surface area
- Due to phonon mismatch across two different solids or phonon mismatch from liquid helium to solid (Kapitza resistance)



# **Very Low T Conductivity and Specific Heat**





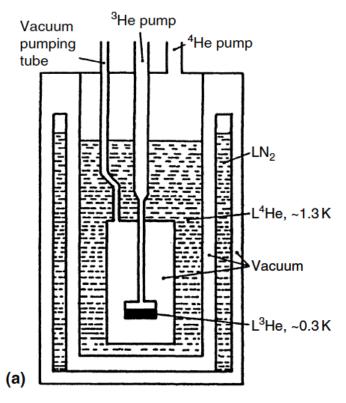
# **Sub Kelvin Refrigeration**

- <sup>3</sup>He sorption coolers
- Dilution refrigerators
- Adiabatic demagnetization



### <sup>3</sup>He Sorption

- Sorption Coolers use a getter to pump the vapor from a liquid reservoir
  - Getter is recycled by heating and the gas is recondensed by a higher temperature stage

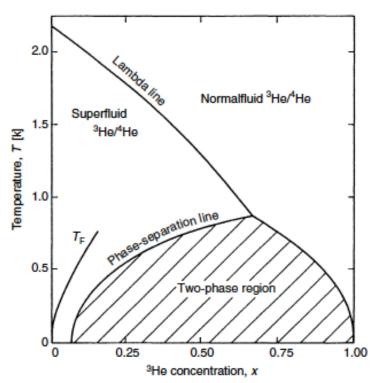


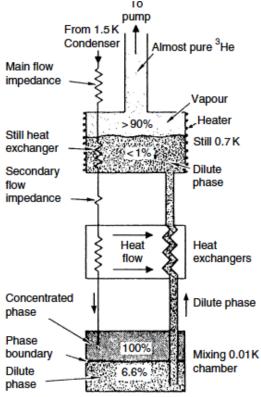


### **Dilution Refrigeration**

- Diluting the lighter isotope <sup>3</sup>He, in liquid <sup>4</sup>He, increases the entropy of the system and therefore cools
- Makes use of the non-zero solubility of <sup>3</sup>He in <sup>4</sup>He even at very low temperatures

Can be made continuous by separating the <sup>3</sup>He out of solution at higher temperature and then re-condensing it

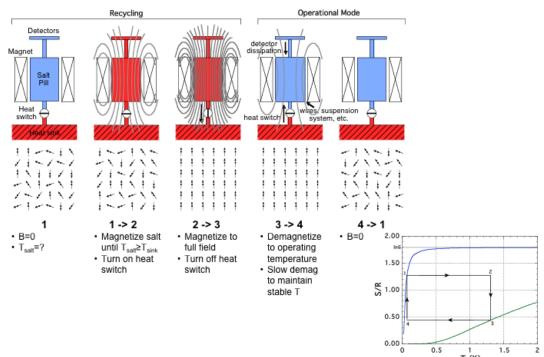






#### **Adiabatic Demagnetization**

- SdT = MdH takes the place of d(ST) = d(PV) in a cryocooler cycle
- [add in ch. 14 material from Zemansky]
- Adiabatic demagnetization refrigeration follows a very Carnot-like cycle of constant S and constant T
  - Produces efficiencies close to Carnot
  - No moving parts for low temperature ADRs using gas-gap heat switches



#### **Continuous ADR**





### **Biliography**

- Spacecraft Thermal Control Handbook, Vol. II
   Cryogenics, Martin Donabedian, Ed., Aerospace Press (2003)
- Cryogenic Engineering, Thomas M. Flynn, CRC Press (2005)



#### **Final Thought**

- It has been said that a problem in low temperature physics can eventually be used to measure and achieve even lower temperatures
  - Problems are actually opportunities!